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A finite element analysis comparison study of the proposed new CEN method for the design of conical shells under combined loadings

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Conical Vessel Design



- EN13445 provides rules for the design of pressurised conical shells based on former East German TGL rules
- These have also been adopted into the UK PD5500 code
- The main shell thickness, e_{con} , is evaluated on the basis of a pressure calculation satisfying simple equilibrium

$$e_{\text{con}} = \frac{P \cdot D_e}{2f \cdot z + P} \cdot \frac{1}{\cos(\alpha)}$$

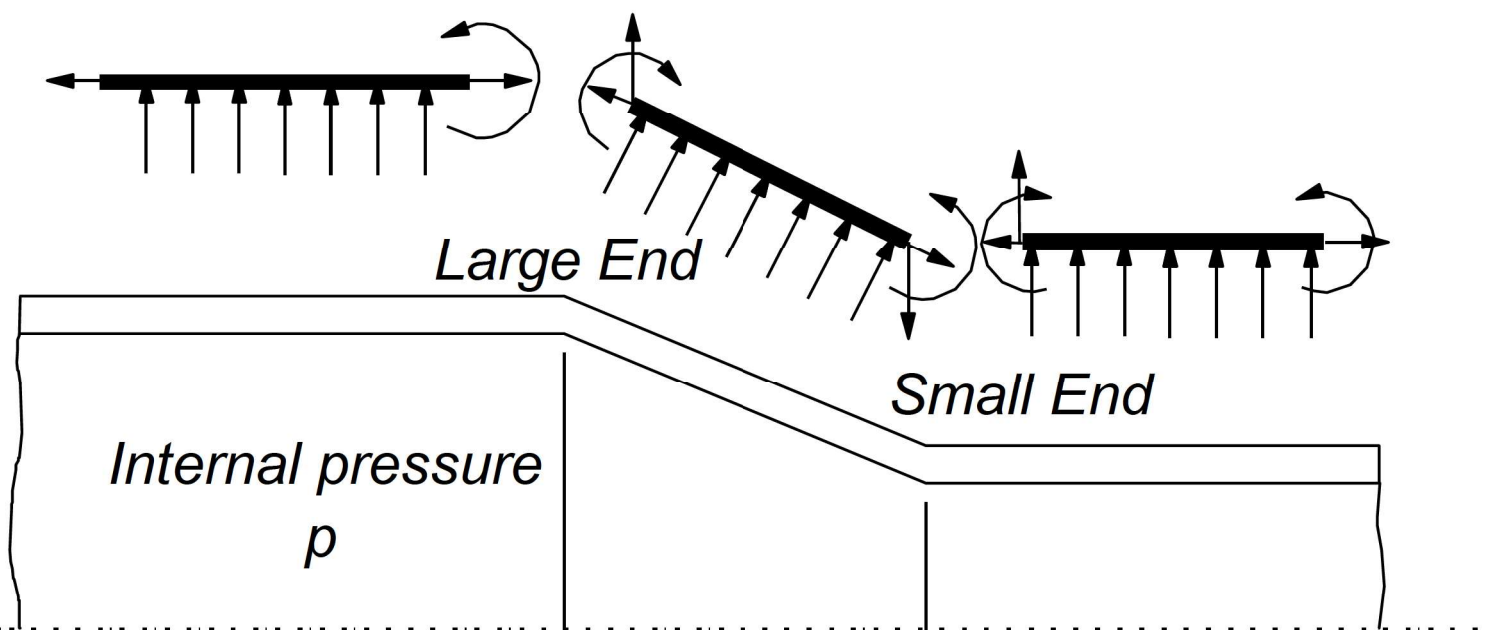
Where P is the internal pressure, D , the diameter, f , the design stress, z , the joint efficiency and α the cone semi-apex angle



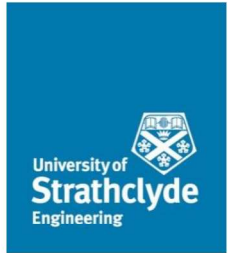


Discontinuity Analysis

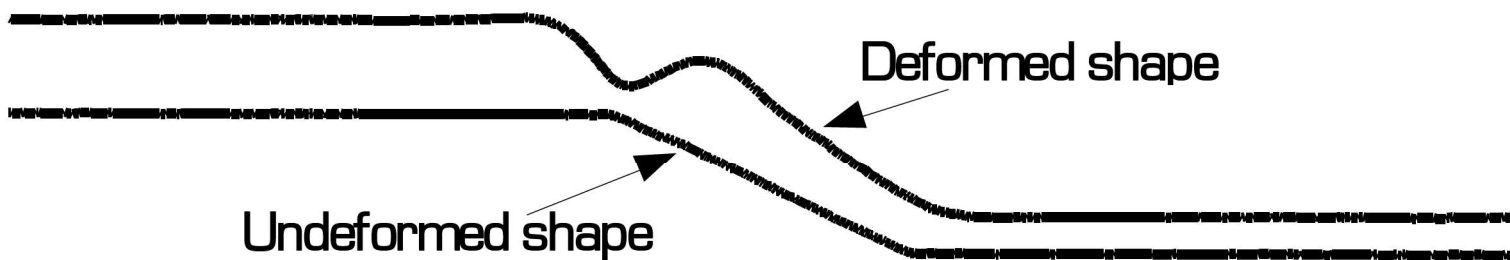
Edge forces resulting in discontinuity stresses



Axi-symmetric FEA

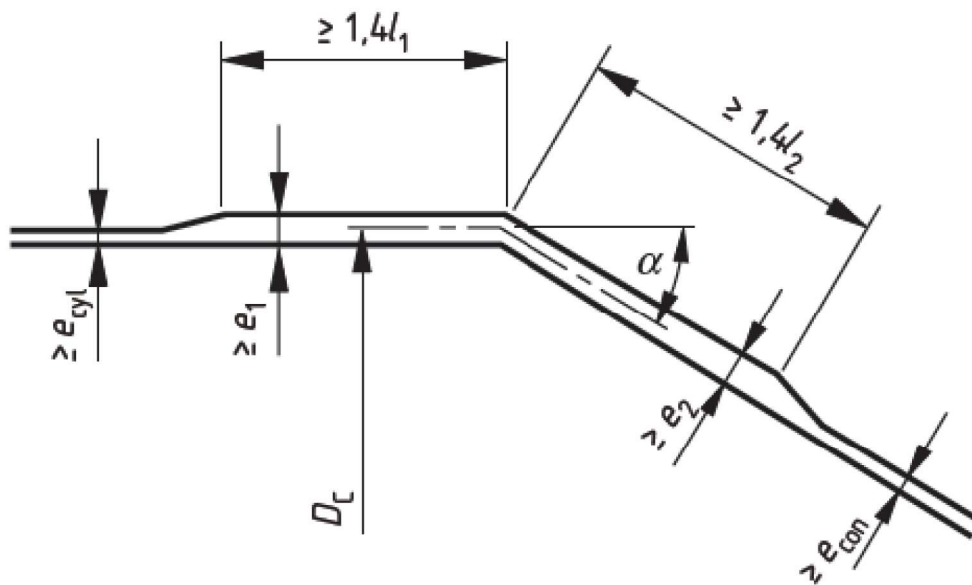


1 An axi-symmetric finite element analysis of the cone/cylinder junctures made using conical shell elements results in the following deformed shape.

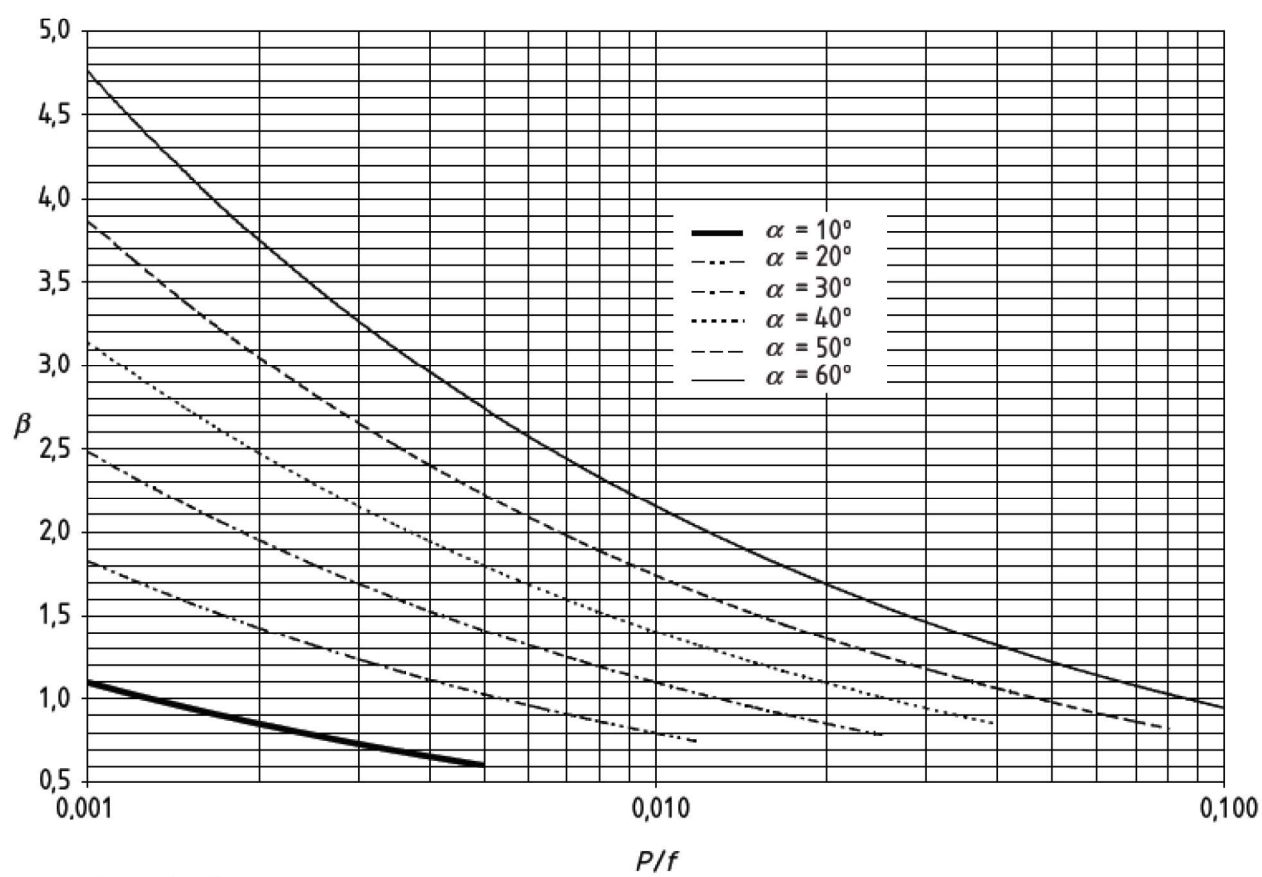


Cone without knuckle

Junction Reinforcement (1)



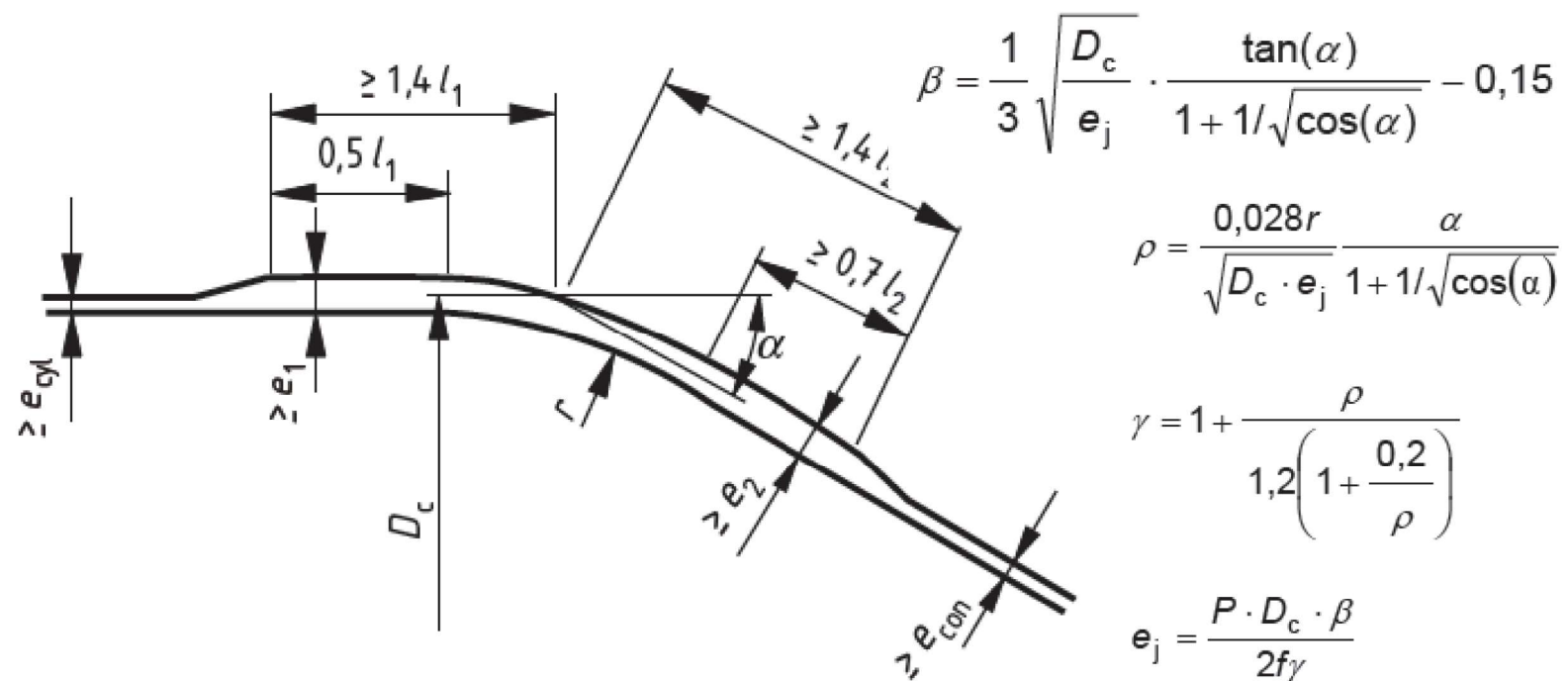
Geometry of cone/cylinder intersection without knuckle –
Large end – EN13445 Figure 7.6-1



$$\beta = \frac{1}{3} \sqrt{\frac{D_c}{e_j}} \cdot \frac{\tan(\alpha)}{1 + 1/\sqrt{\cos(\alpha)}} - 0,15$$

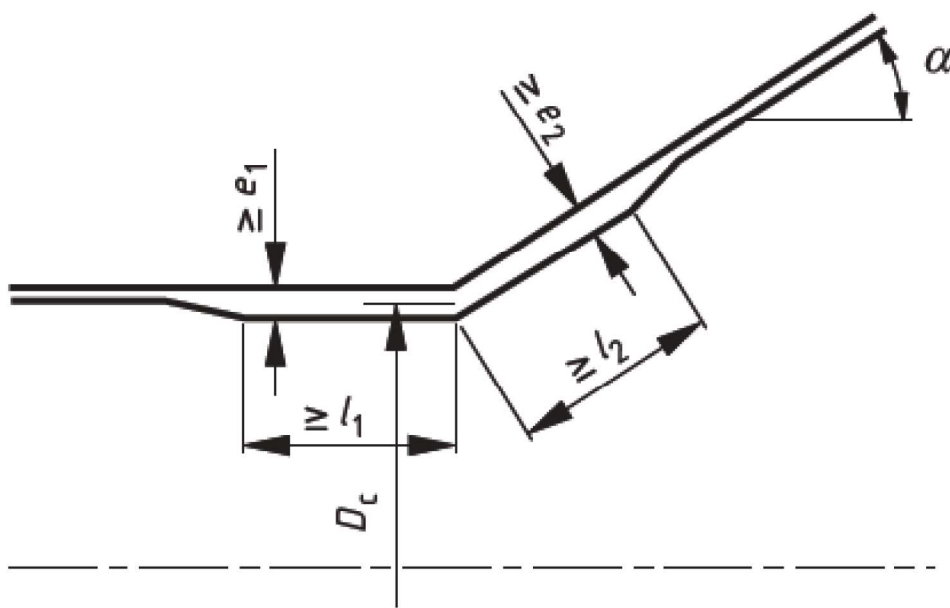
$$e_j = \frac{P \cdot D_c \cdot \beta}{2f}$$

Junction Reinforcement (2)



Geometry of cone/cylinder intersection with knuckle –
Large end – EN13445 Figure 7.6-2

Junction Reinforcement (3)



Geometry of cone/cylinder intersection –
Small end – EN13445 Figure 7.6-3

$$s = \frac{e_2}{e_1}$$

when $s < 1$

$$\tau = s \sqrt{\frac{s}{\cos(\alpha)}} + \sqrt{\frac{1+s^2}{2}}$$

when $s \geq 1$

$$\tau = 1 + \sqrt{s \left\{ \frac{1+s^2}{2\cos(\alpha)} \right\}}$$

$$\beta_H = 0,4 \sqrt{\frac{D_c}{e_1}} \cdot \frac{\tan(\alpha)}{\tau} + 0,5$$

If

$$P \leq \frac{2f \cdot z \cdot e_1}{D_c \cdot \beta_H}$$

Limitations



- Semi-apex angle greater than 75degrees
- Cones for which

$$\frac{e_a \cdot \cos(\alpha)}{D_c} \leq 0,001;$$

- Short cones joining to a jacket
- Limitations on each case for die-out distances
- NO method for global loads –
 - thrust force
 - overturning moment
 - Interaction with pressure loading

Global Loads – EN13445 Clause 16.14

- Rules are given for determining the minimum thickness of a cylindrical shell subject to a combination of loads in addition to pressure, at sections remote from the area of application of local loads and from structural discontinuities.
- Simple equilibrium calculations
- Simple interaction relationships
- Checks for maximum compressive stress
- Permissible compressive stress check
- Tolerance checks – out-of-roundness
- Caution over wind and earthquake regions

Kiesewetter Method – new 16.15

- Proposed new method – KM Method
- Limit analysis based on the formation of plastic hinges
- Postulated deformation modes
- Two calculation routes –
 - Can be considered as a ‘conical transition
 - Otherwise large and small ends considered seperately
- Loads applicable for FEA study

$$P = 0.3 \times \frac{2e_c f_c \cos \alpha}{D_L}$$

$$M_{BS} = 0.3 \times \frac{\pi D_s^2 e_s f_s}{4}$$

$$F = -0.3 \times \pi D_s e_s f_s$$

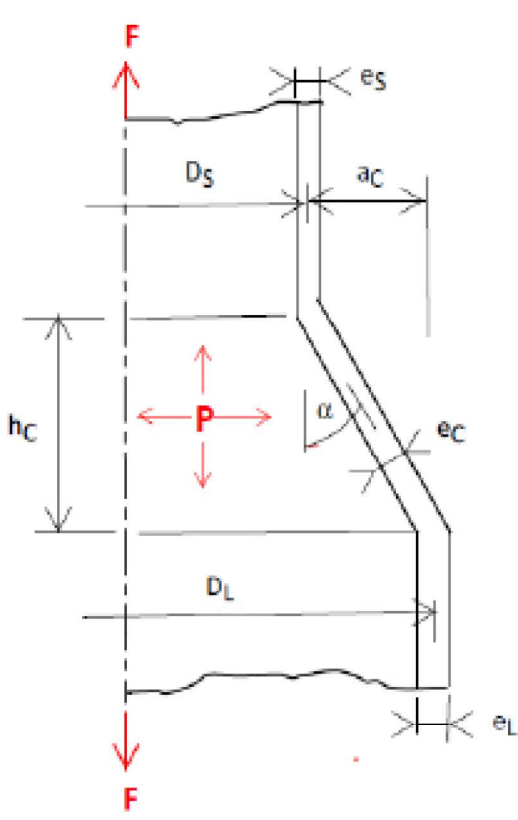


Figure 2a: Sketch of conical transition loaded by int. pressure and axial force

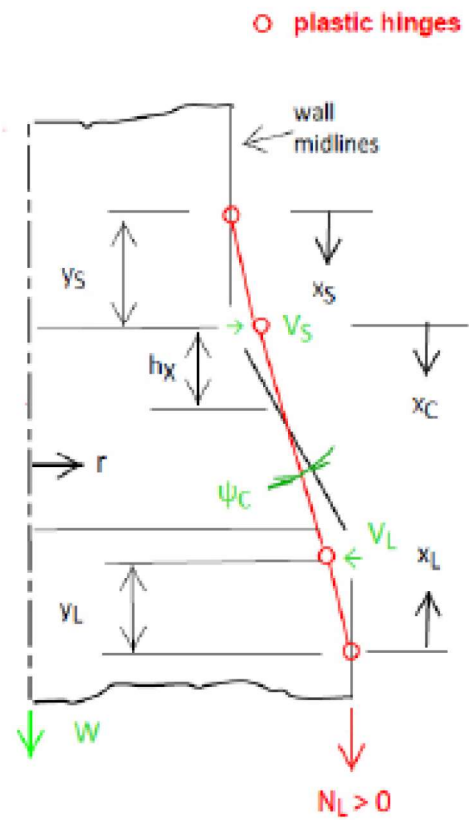


Figure 2b: Deformed model for resulting axial **tension** force

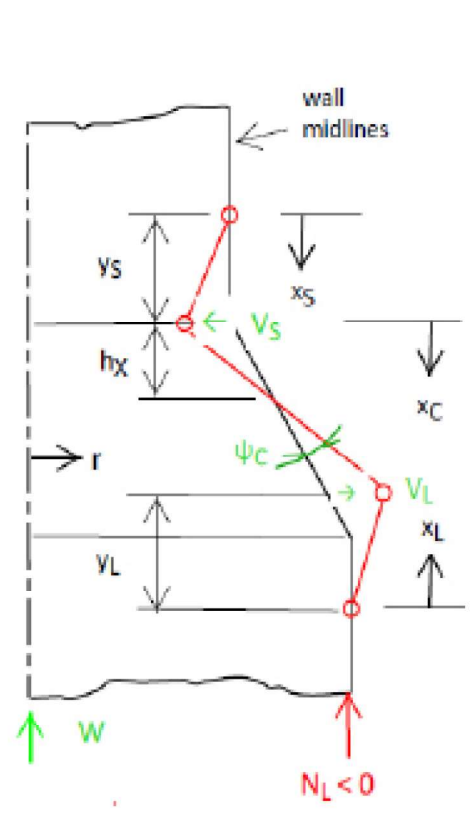


Figure 2c: Deformed model for resulting axial **compression** force

EN 13445 Class
NOT fulfilled
(Alternative design
pressure case)

The resultant
 $\max\{-(Z_s + F_{ps}) ; -Z_{ss} ; -Z_{cs}\}$

The resultant

Justification
(S)

DBF of junction between conical shell and cylindrical shell

DBF of junction between cylindrical
shell and conical shell at small end of
the cone.
(Kiesewetter Method)

Calculate:
 δ and γ

Calculate:
 $\sin \alpha$, f_{sc} , f_{ss} , f_{uc} , f_{uc} , and Φ
where radicands for f must be greater than or
equal to zero.

Calculate:
 Y , q , D_T , and ρ .

Calculate:
 Z_{ss} and Z_{cs}
where radicands must be greater than or equal
to zero.

Calculate:
 Z_s and F_{ps}

The resultant (effective) axial force F_{ps} on the small end of the conical shell shall be checked by the following
condition:

$$\max\{-(Z_s + F_{ps}) ; -Z_{ss} ; -Z_{cs}\} \leq F_{ps} \leq \min\{(Z_s - F_{ps}) ; Z_{ss} ; Z_{cs}\}$$

If condition is not met:
Small end junction is
NOT acceptable.

If condition is met:
Small end junction is
acceptable.

The design of the cylindrical shell and the conical shell at the
small end under global loads is acceptable.

Parametric Range



Cone Angle (°)	e/D	D_s/D_L
15	0.002	0.2
30	0.005	0.3
45	0.01	0.45
60	0.02	0.65
75	0.05	0.8

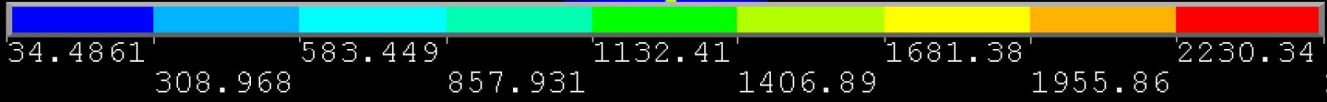
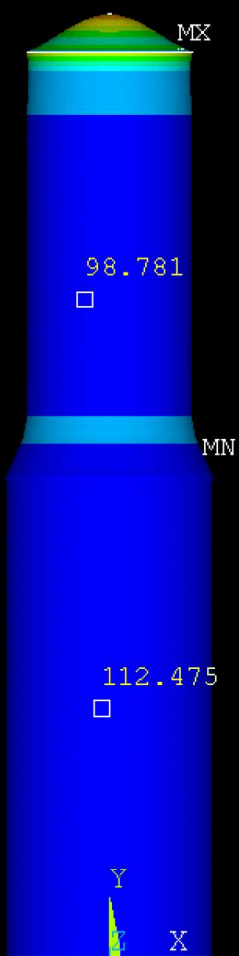
Note: D_s/D_L is the ratio of small end to large end and e/D is 0.002 to 0.16.

NODAL SOLUTION

STEP=1
SUB =1
TIME=1
SINT (AVG)
DMX =32.5262
SMN =34.4861
SMX =2504.82

ANSYS R17.1
Academic

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1

NODAL SOLUTION

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SUB =1

TIME=1

SINT (AVG)

DMX =5.35025

SMN =.257E-03

SMX =111.053

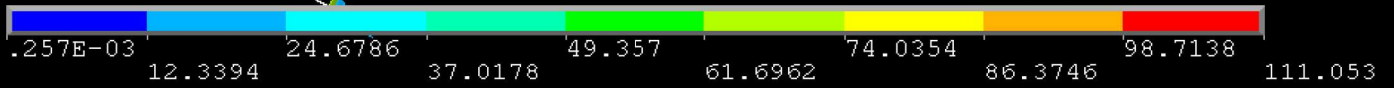
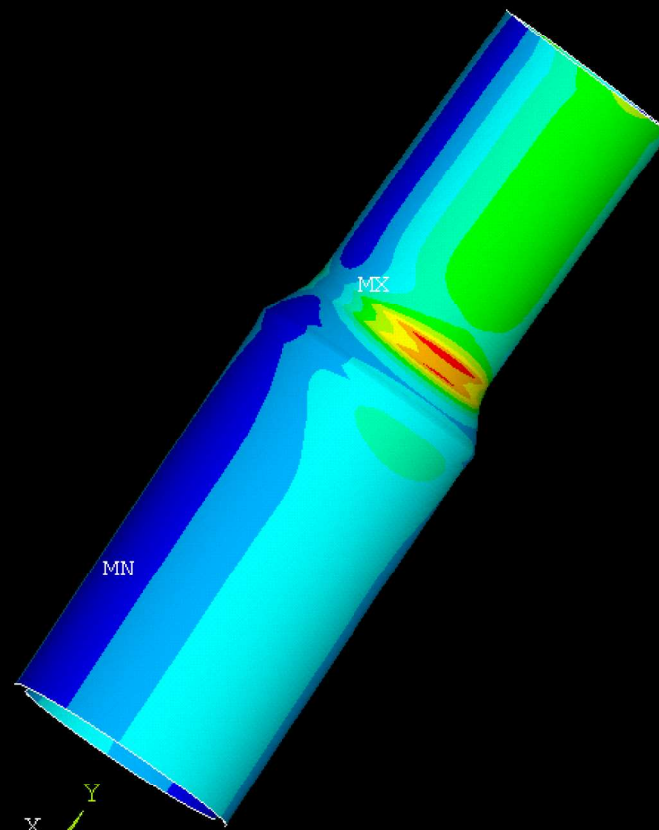
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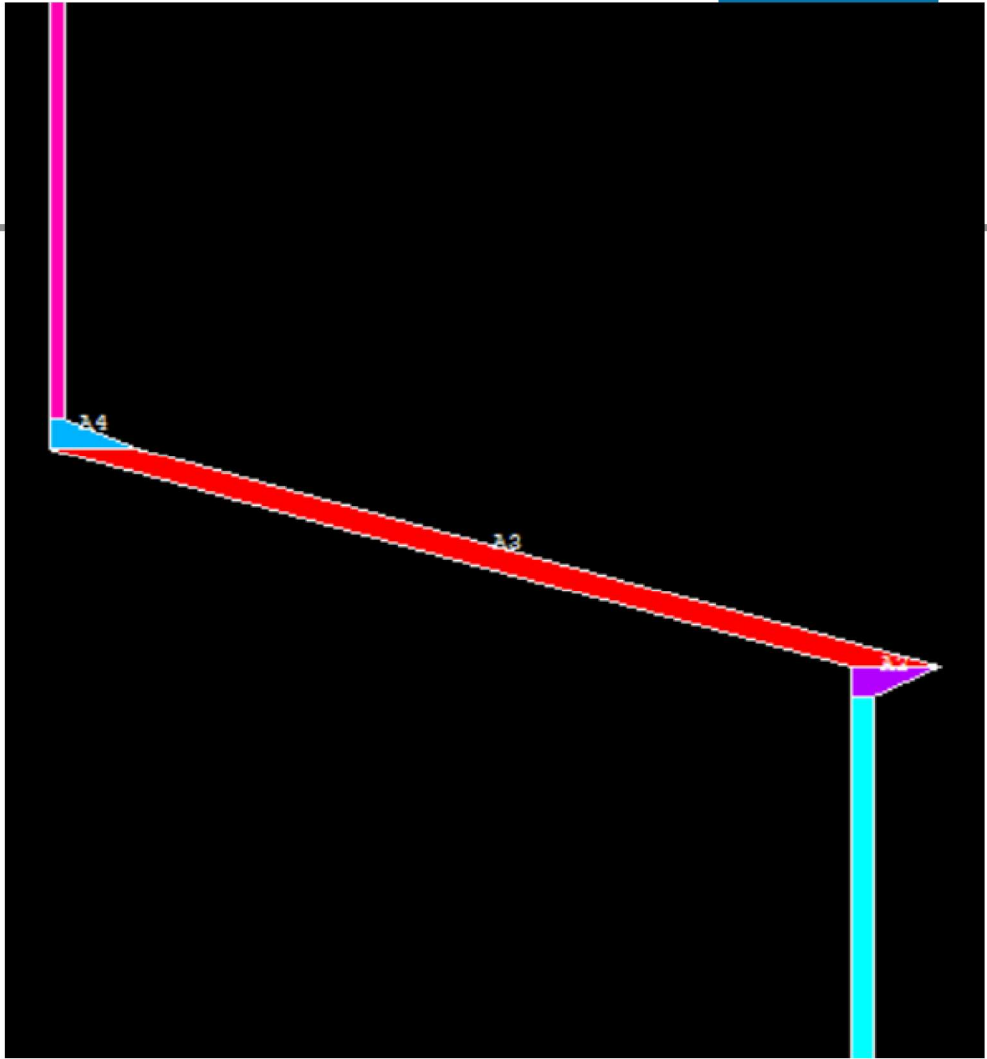
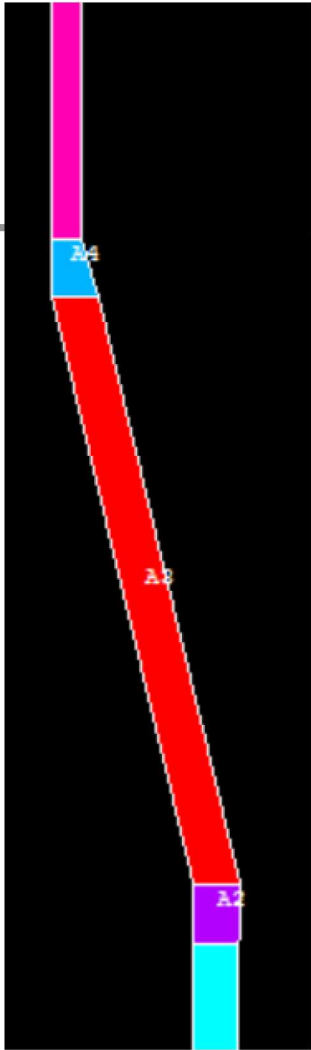
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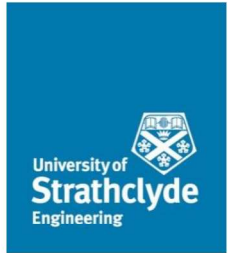
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Conical Shell Analysis



Pass/Fail Criterion

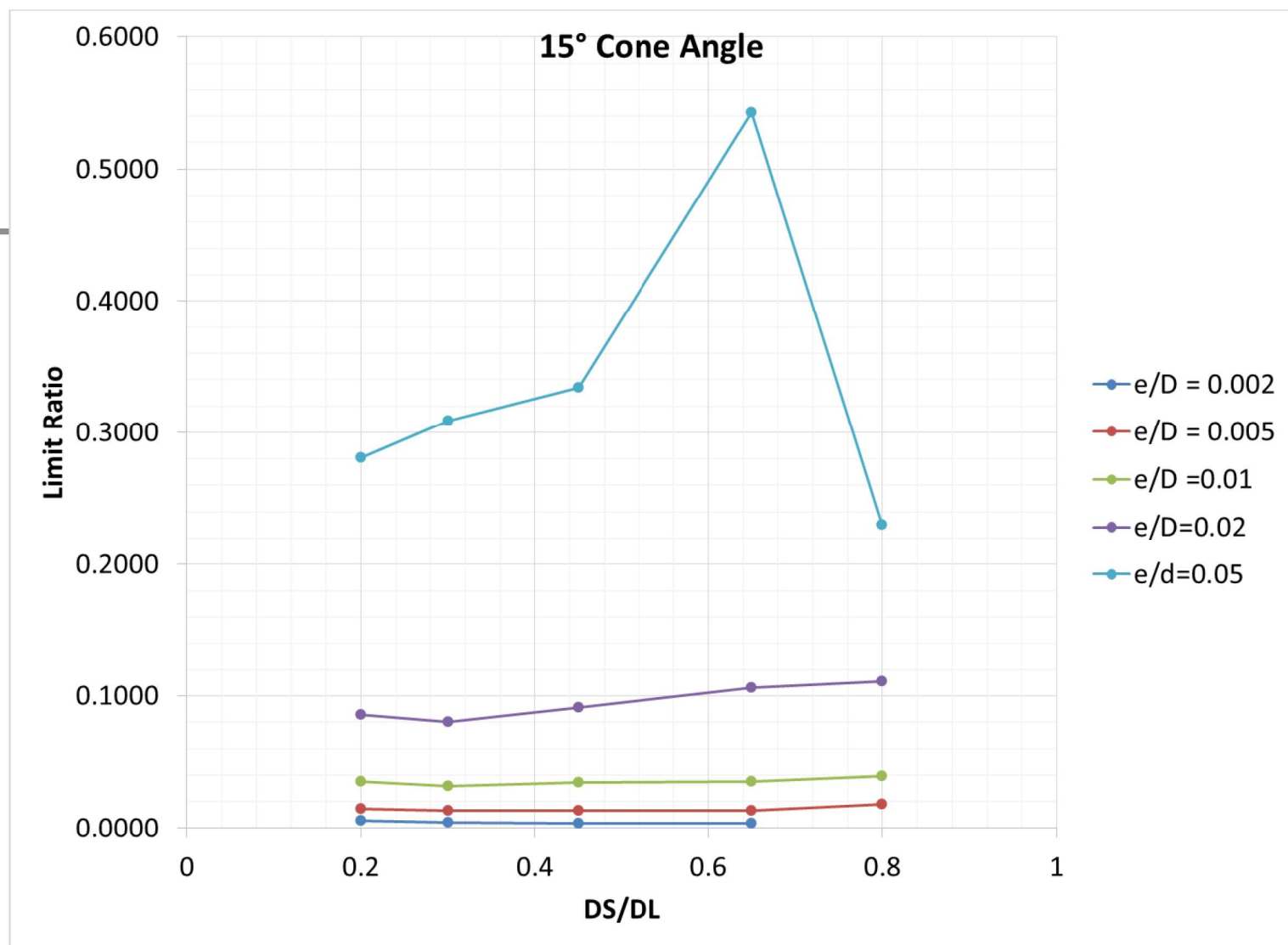


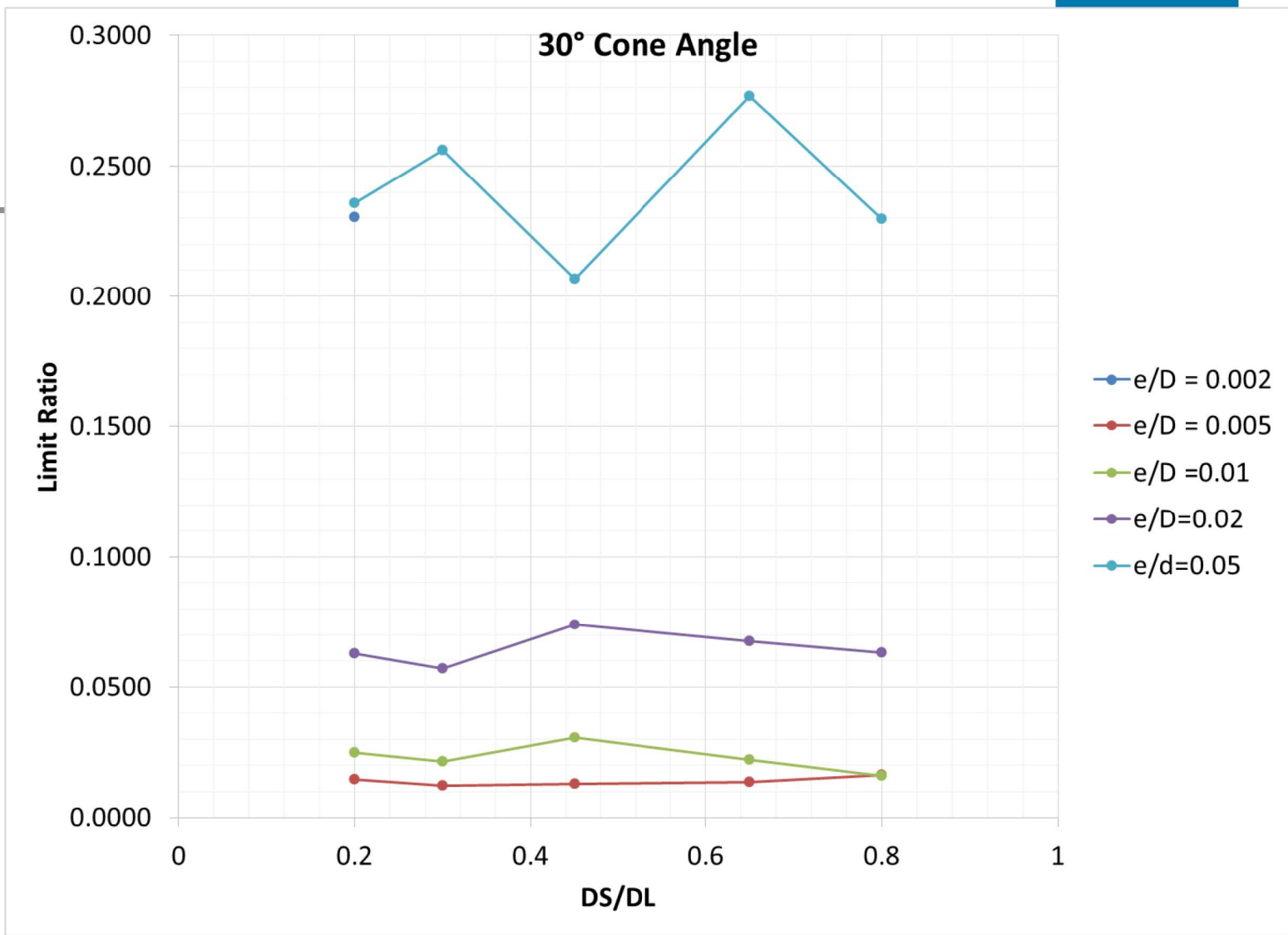
$$\textit{Proportion of Allowable Stress} = \frac{\textit{FEA Result}}{f}$$

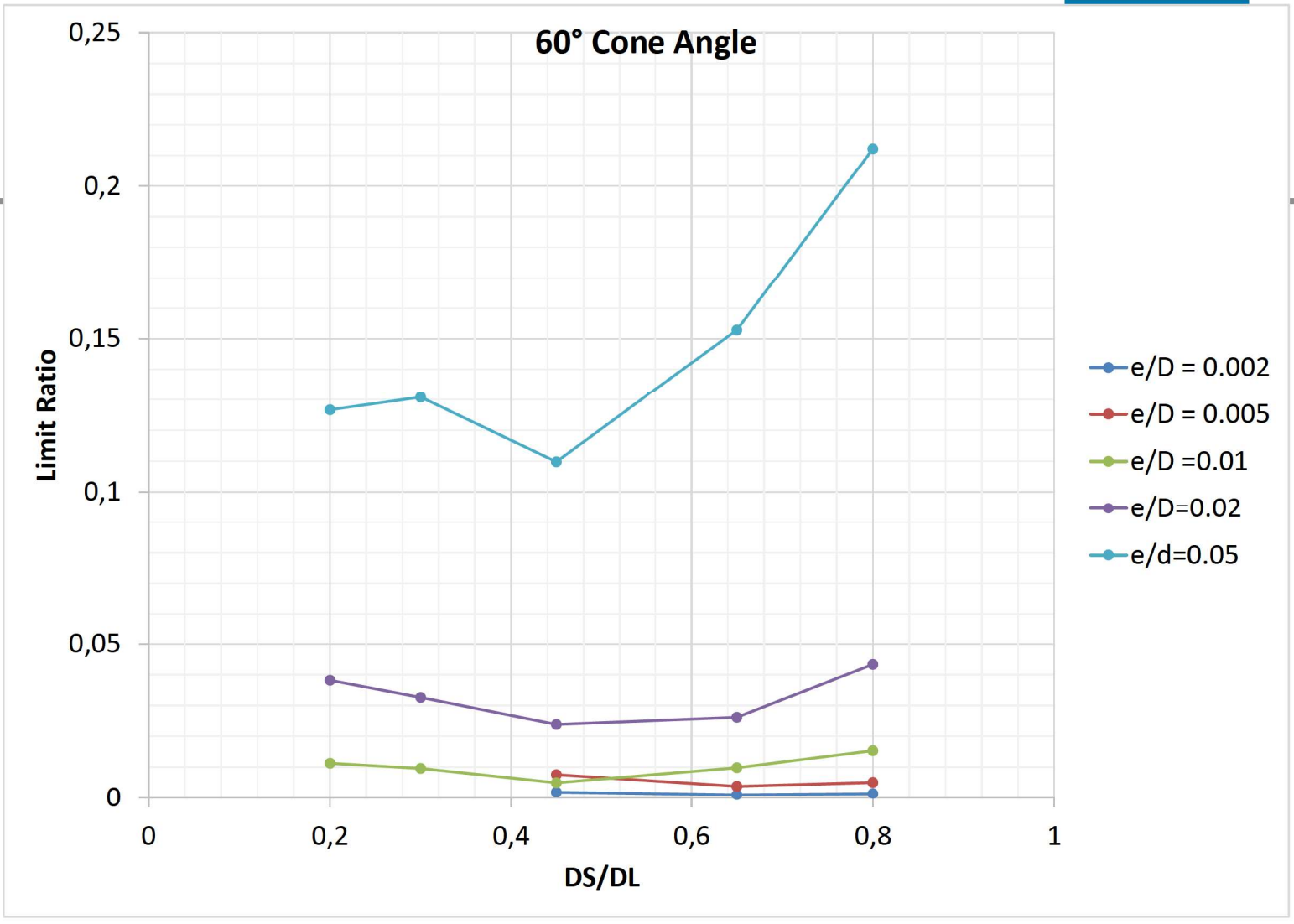
$$\max\{-(Z_T + F_{PC}) ; -Z_{SS} ; -Z_{LL} ; -Z_{CS} ; -Z_{CL}\} \leq F_{RC} \leq \min\{(Z_T - F_{PC}) ; Z_{SS} ; Z_{LL} ; Z_{CS} ; Z_{CL}\}$$

$$\textit{Proportion of Limit Load} = \frac{F_{RC}}{\textit{Limit Load}}$$

$$\textit{Limit Ratio} = \frac{\textit{Proportion of Limit Load}}{\textit{Proportion of Allowable Stress}}$$







Discussion & Conclusions



- KM method has been implemented in Excel and FEA analysed
- There are modelling issues around the junctions cf. KM however for 'regular vessels' benchmark case is validated
- KM/FEA do not represent the junctions well – and excessively high stresses result (can be fixed – but not with KM)
- Some parameter ratios should be withdrawn as they represent extreme cases (e.g. 75° case)
- Some thickness to diameter ratios also produce unrealistic relationships – excessively thick and thin configurations
- More work needed on representative parametric range
- Future limit analysis to be undertaken

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Thank You!

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